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Patentanmeldung Nr. Patent application No. Demande de brevet n°

05251424.7

Der Präsident des Europäischen Patentamts;
Im Auftrag

For the President of the European Patent Office

Le Président de l'Office européen des brevets
p.o.

R C van Dijk



Anmeldung Nr:
Application no.: 05251424.7
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Anmelder/Applicant(s)/Demandeur(s):

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Bezeichnung der Erfindung/Title of the invention/Titre de l'invention:
(Falls die Bezeichnung der Erfindung nicht angegeben ist, siehe Beschreibung.
If no title is shown please refer to the description.
Si aucun titre n'est indiqué se referer à la description.)

Tow cable termination

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TOW CABLE TERMINATION

This invention relates to a tow cable termination for sustaining the tensile load when an object is towed at high speeds at the end of a cable, in particular to a termination used to connect a tow cable to an electronic warfare decoy used in fast jet applications for protection against missile attack.

One technique used to protect jet aircraft against missile attack is an electronic decoy, which is towed behind the aircraft by a cable. Due to the high performance of these aircraft and the significant trailing vortices created by the aircraft wings, particularly during high "G" manoeuvres, the method by which the 10 cable is terminated and connected to the decoy is fundamental to the successful operation of the towed decoy. The cable termination has to transmit those forces acting on the decoy in flight (aerodynamic drag and inertia) into tow cable.

The decoy's flight performance follows from its aerodynamic design, 15 which to a large extent is determined by the relative positions of the centres of pressure and mass of the decoy. With a highly stable decoy, the decoy's centre of mass follows the tow cable, so that the cable termination transmits the loads from the decoy to the cable with little off-axis deviation. However, with a less stable decoy, or in a very high vortex environment or during high "G" 20 manoeuvres, the axis of the decoy can fluctuate or deviate significantly with respect to the axis of the cable, typically up to a cone angle of plus or minus 90 degrees. In these circumstances, either the cable termination has to flex over this extremely wide angle of movement, or the tip of the tow cable itself has to flex over this range. Modern materials such as Kevlar (Kevlar is a trademark of 25 E.I. DuPont de Nemours and Company) have a very high strength to weight ratio and therefore are ideal for use as tow cables. However, even such modern materials cannot sustain local bending in a highly dynamic environment, and the rapid and persistent flexure of the cable near to the cable termination in a high vortex environment can lead to catastrophic failure. Specialist cable 30 terminations have been proposed, but these generally have a limited angle of articulation, typically less than a cone angle of plus or minus 30 degrees. Techniques to relieve cable strain, to compensate for this low articulation, have

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been used, but these methods are insufficiently robust significantly to extend the life of the cable/cable termination arrangement. A low angle of articulation at the cable termination results in the tow cable flexing near the tip of the cable, resulting in the failure of the tow cable and the loss of the decoy. This mode of failure places a severe restriction on the flight envelope of the aircraft when 5 towing a decoy.

Accordingly, the present invention provides a tow cable termination for sustaining the tensile load when an object is towed at high speeds at the end of a cable, the termination comprising a closed eye bolt attached to the cable and 10 a U-bolt looped through the eye bolt and attached to the object.

Such an arrangement enables a high degree of articulation, up to a cone angle of plus or minus 90 degrees, whilst easily being capable of sustaining the normal tensile loads sustained when a fast jet aircraft is towing a decoy and of transmitting these loads into the tow cable without also transmitting any of the 16 lateral forces which would cause the cable to flex.

Preferably the U-bolt is attached to the object so as to protrude from the object. This is an important factor in enabling the articulation; because the linkage is outside the envelope of the towed object (typically a decoy radome) the motion of the linkage is not restricted by the object itself impinging on the 20 linkage.

The cable termination may comprise a cable barrel mounted to the end of the cable, the eye bolt being releasably attachable to the cable barrel. The cable barrel is the main interface to the tow cable. Within this component there may be features which enable the cable to be fixed by a conventional resin 25 potting process so as to form a joint having a strength equivalent to that of the tow cable. Another desirable feature of the cable barrel is that it enables the optical and electrical conductors to be split out of the cable structure and routed to the decoy for connection to the electronics systems therein. In this case it is advantageous for the eye bolt to have an aperture for allowing at least part of 30 the cable to extend through to connect to the object. Thus, the electronic and

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other systems in a decoy can be connected to the aircraft without the connection having to sustain any tensile loads caused by towing the decoy.

The cable barrel and the eye bolt are preferably provided with matching screw threads, so as to be releasably attachable. The aperture in the eye bolt 5 allows optical and/or electrical conductors running down the core of the cable to connect through to the tow decoy. Due to the high dynamic motion and loads experienced on the eye bolt and the U-bolt where these are now looped together, one or both are preferably coated with a low friction material (such as by electroplating with a low friction coating) to enable the two elements to move 10 freely as they articulate (pivot) in flight.

The U-bolt is preferably adapted to connect to an interface piece within the towed object so as to connect the U-bolt to the object and to permit pivotable motion therebetween. The design of the U-bolt and the interface piece is such as to minimise bending moments and fatigue stresses. The 15 interface piece provides mechanical support to the U-bolt, by ensuring the structure is closed around the two arms of the U-bolt to the greatest extent possible whilst still maintaining the required kinematic motion. It is also important to maintain the minimum length of U-bolt commensurate with the degree of articulation required. Accordingly the U-bolt may be threaded at the 20 ends and attached to the object by nuts threaded thereonto. This arrangement allows the length of U-bolt protruding from the object to be adjusted easily. Preferably the exterior of the interface and the interior of the object have matching sliding surfaces, such as spherical surfaces, to permit sliding therebetween, thus permitting a certain amount of pivotable motion whilst 25 providing a firm mounting. Integral with the interface between the object and the interface piece may be environment seals to prevent the ingress of water and dust.

The invention will now be described by way of example only and with reference to the accompanying Figure 1, which shows an exploded view of a 30 tow cable termination assembly in accordance with the present invention.

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The assembly shown in Figure 1 comprises four main components for connection to the tow cable (not shown, but in use would be on the left hand side of the assembly shown in the drawing). The first component is the cable barrel 1 which is the main interface to the tow cable (not shown). The cable barrel 1 is in the form of a hollow cylinder which is connected to the cable by a convention resin potting process, forming a joint having a strength equivalent to that of the cable itself. Part of the exterior surface of the cable barrel 1 (the right hand portion as shown in the Figure) has a screw thread adapted to engage with a matching screw thread on the interior of eye bolt 3.

Eye bolt 3 is part of the articulation linkage; it is a high strength component and has an internal screw thread to accept the cable barrel 1. At one end (to the right as shown in the drawing) the eye bolt has a closed link (preferably circular) which mates with the U-bolt 5. Eye bolt 3 has an aperture (not shown) allowing optical and electrical conductors in the cable (not shown) to exit the eye bolt for connecting with the systems internally of the object to be towed.

The eye bolt 5 is secured to the object to be towed (illustrated here by radome 9) by interface piece 7 and nuts 11. The legs of U-bolt 5 are threaded for engagement with nuts 11, which hold interface piece 7 against the interior surface of radome 9. Interface piece 7 has spherical exterior surface adapted to engage with a matching surface on the inside of radome 9 so as to allow a certain amount of sliding movement so that the radome can pivot (although the main articulation takes place where the U-bolt 5 is looped through eye bolt 3).

In use, as the cable and cable barrel 1 are towed through the air at fast jet speeds and the towed object 9 is buffeted and forced to oscillate by the trailing vortices from the towing aircraft, the towed object 9 pivots relative to the eye bolt 3 thus minimising the amount of stress applied to the tow cable itself.

Although described in connection with an object being towed through the air, the principles of the invention apply equally to objects being towed through water, for example to sonar decoys.

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Claims

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1. A tow cable termination for sustaining the tensile load when an object is towed at high speeds at the end of a cable, the termination comprising a closed eye bolt attached to the cable and a U-bolt looped through the eye bolt and attached to the object.
5
2. A termination as claimed in claim 1 wherein the U-bolt is attached to the object so as to protrude from the object.
3. A termination as claimed in claim 1 or 2 comprising a cable barrel mounted to the end of the cable, the eye bolt being releasably attachable to the cable barrel.
10
4. A termination as claimed in claim 1, 2 or 3 wherein the eye bolt has an aperture for allowing at least part of the cable to extend through to connect to the object.
5. A termination as claimed in any preceding claim comprising an interface piece adapted to connect the U-bolt to the object and to permit pivotal motion therebetween.
15
6. A termination as claimed in claim 5 wherein the exterior of the interface piece and the interior of the object have matching spherical sliding surfaces which permit sliding therebetween.
- 20 7. A termination as claimed in claim 5 or 6 comprising seals between the object and the interface piece for preventing water and/or dust ingress.
8. A termination as claimed in any preceding claim wherein the U-bolt is threaded at the ends and attached to the object by nuts threaded there onto.
- 25 9. A termination as claimed in any preceding claim wherein the eye bolt and/or the U-bolt is/are coated with a low friction material.

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ABSTRACT

A specialist tow cable termination for sustaining the tensile load when an object is towed at high speeds at the end of a cable behind an aircraft, the termination comprises a closed eye bolt attached to the cable and a U-bolt 5 looped through the eye bolt attached to the object. The arrangement allows significant pivoting bending stresses to the tow cable.

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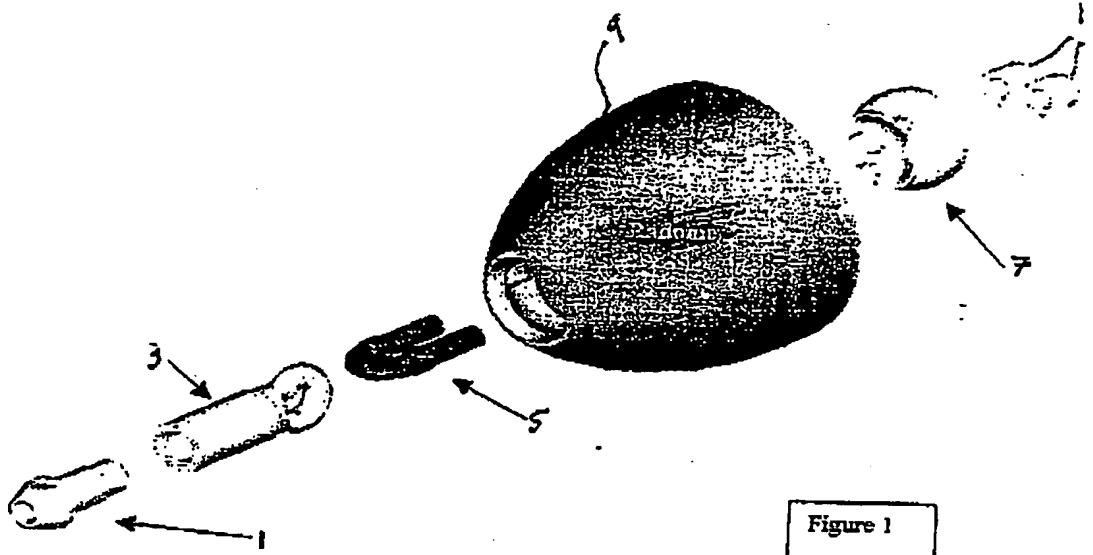


Figure 1

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OF PRIORITY DOCUMENT

(PCT Administrative Instructions, Section 411)

Date of mailing (day/month/year) 17 May 2006 (17.05.2006)	To: BAE SYSTEMS plc, Group IP Dept P.O. Box 87 Lancaster House Farnborough Aerospace Centre Farnborough Hampshire GU14 6YU ROYAUME-UNI
Applicant's or agent's file reference XA2108	IMPORTANT NOTIFICATION
International application No. PCT/GB2006/050046	International filing date (day/month/year) 03 March 2006 (03.03.2006)
International publication date (day/month/year) Not yet published	Priority date (day/month/year) 09 March 2005 (09.03.2005)
Applicant SELEX SENSORS AND AIRBORNE SYSTEMS LIMITED et al	

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<u>Priority date</u>	<u>Priority application No.</u>	<u>Country or regional Office or PCT receiving Office</u>	<u>Date of receipt of priority document</u>
09 March 2005 (09.03.2005)	0504842.6	GB	21 March 2006 (21.03.2006)
09 March 2005 (09.03.2005)	05251424.7	EP	08 May 2006 (08.05.2006)

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Date of mailing (day/month/year) 01 May 2006 (01.05.2006)	To: BAE SYSTEMS plc, Group IP Dept P.O. Box 87 Lancaster House Farnborough Aerospace Centre Farnborough Hampshire GU14 6YU ROYAUME-UNI
Applicant's or agent's file reference XA2108	IMPORTANT NOTIFICATION
International application No. PCT/GB2006/050046	International filing date (day/month/year) 03 March 2006 (03.03.2006)
International publication date (day/month/year) Not yet published	Priority date (day/month/year) 09 March 2005 (09.03.2005)
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